

# Flight Test Safety Workshop

## 27-30 April 2009

### Ottawa, Ontario, Canada



# **777 Freighter**

## **Aileron Vibration Occurrence**

Flight Test Safety Workshop  
Ottawa, Ontario, Canada  
April 29-30, 2009

Van Chaney, Boeing 777 Deputy Chief Pilot

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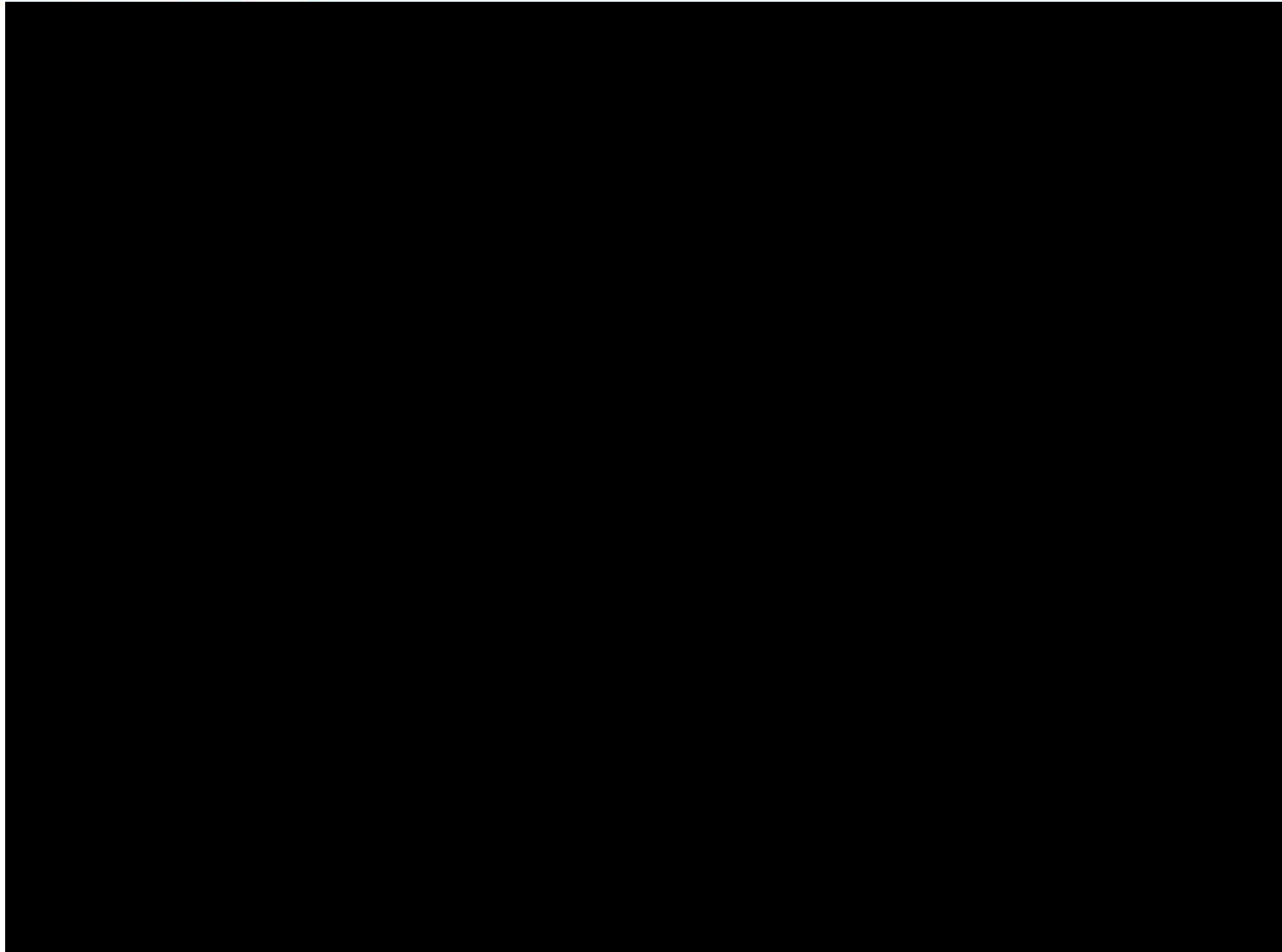
# Agenda

- **Introductions**
- **777F airplane description/background**
- **Original flight test plans**
- **Aileron vibration discovery flights**
- **To fix or not to fix ?**
- **Re-test flights**
- **Lessons Learned**
- **Summary**

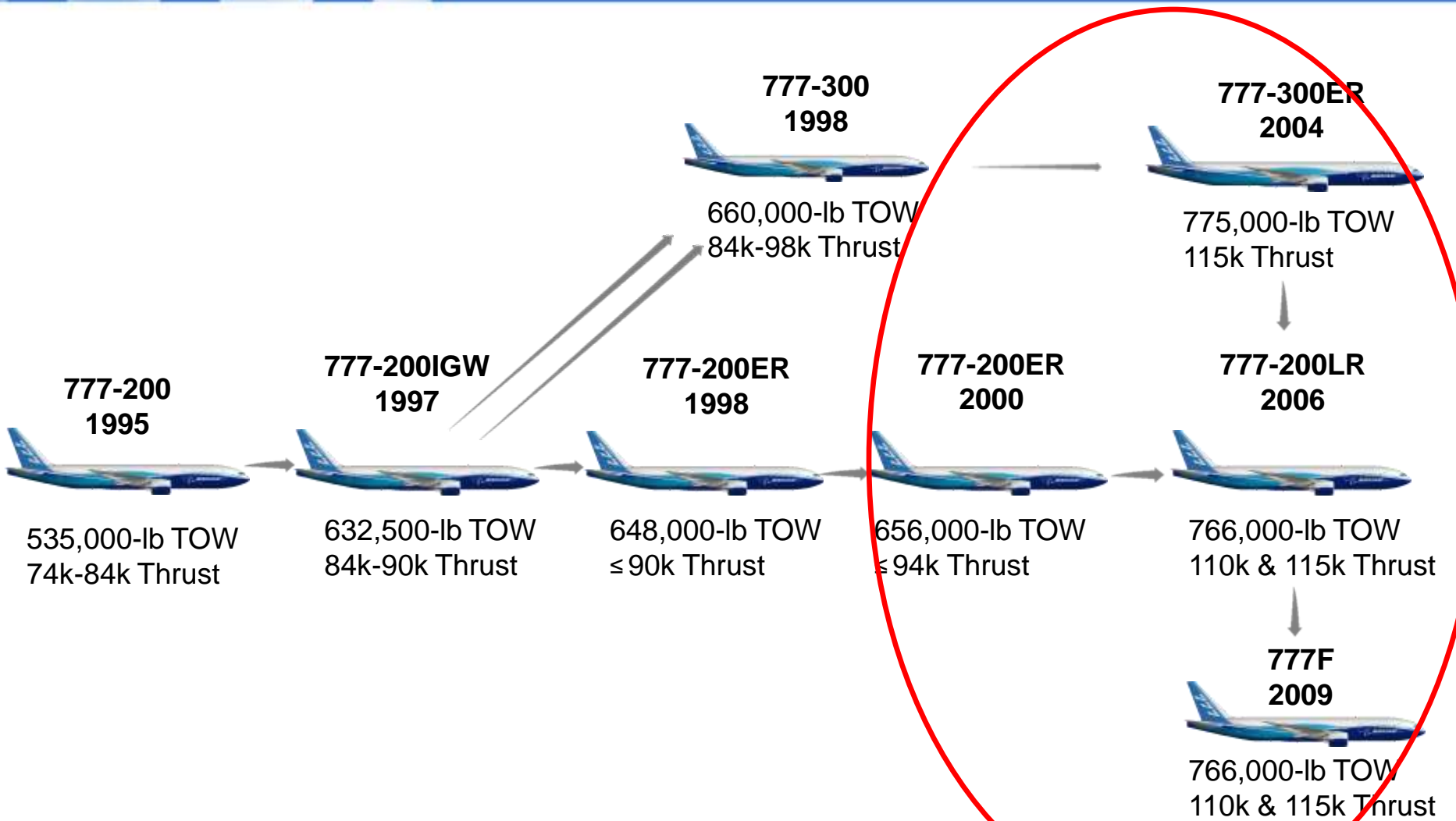
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# 777 Freighter First Flight



# 777 Introduction Family Evolution



# Summary of Changes: 777-200LR to 777F

**MTOW** 766,800 lb (*with revised c.g. envelope*)  
**MLW** 575,000 lb (a 17% increase)

## No aerodynamic changes

- Same wing
- Same fuselage
- Same empennage



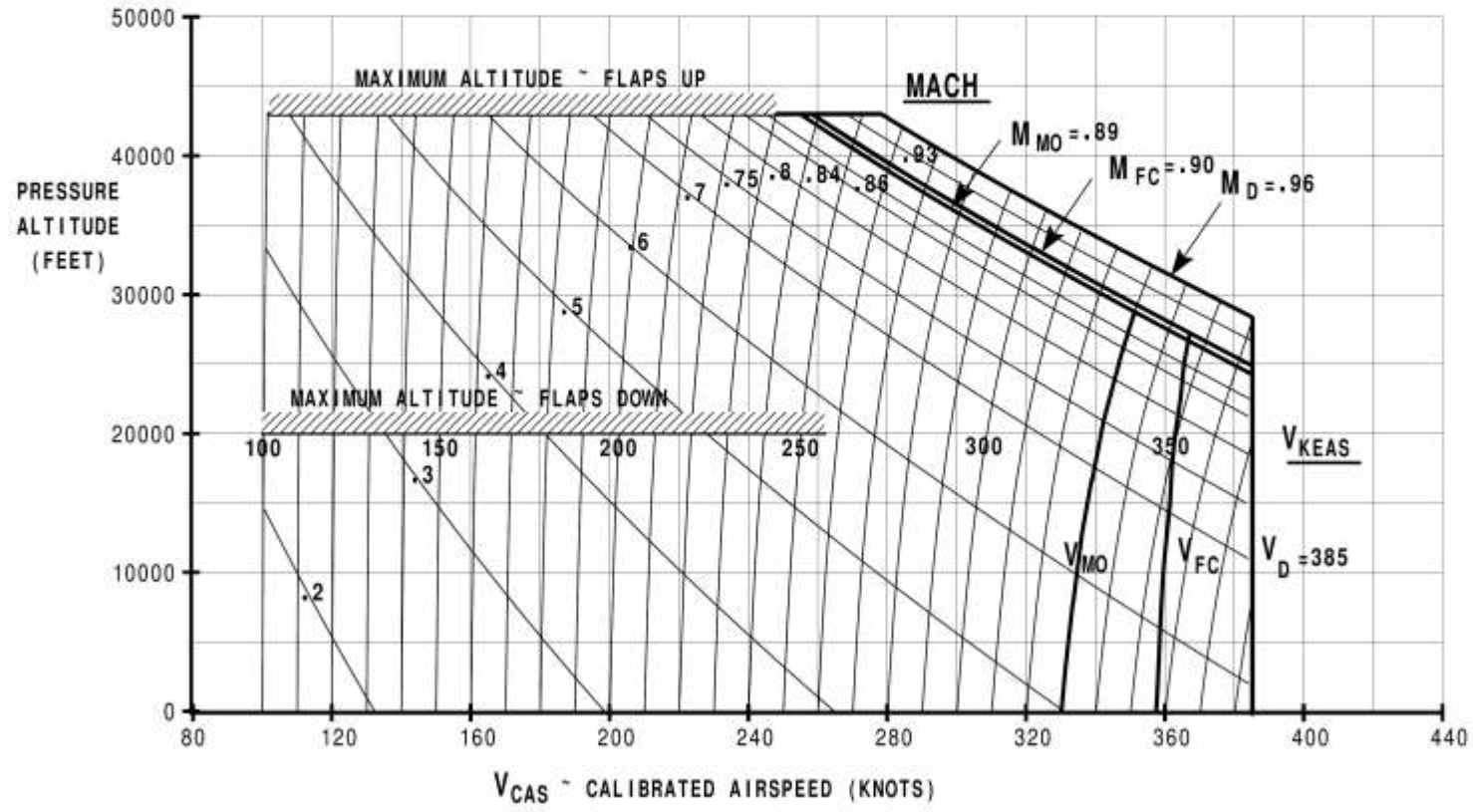
*Addition of Maneuver Load Alleviation (MLA) function in Primary Flight Computers*

# 777F Airplane Description Flight Envelope

- $V_{MO} / M_{MO}$  ~ MAXIMUM OPERATING LIMIT SPEED
- $V_{FC} / M_{FC}$  ~ MAXIMUM SPEED FOR STABILITY CHARACTERISTICS
- $V_D / M_D$  ~ DESIGN DIVE SPEED
- $V_{LO} / V_{LE} = 270$  KEAS ~ GEAR OPERATING SPEED
- $M_{LO} / M_{LE} = .82$  ~ GEAR OPERATING MACH NUMBER

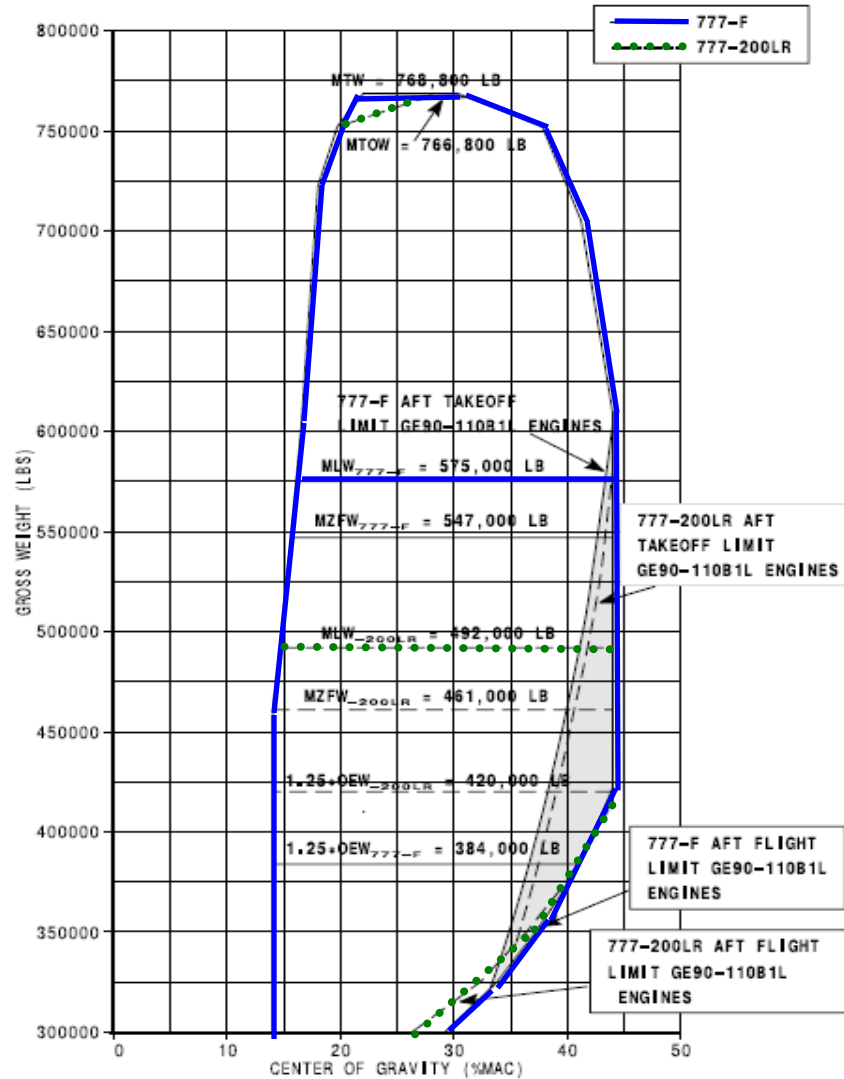
$V_{FE}^*$	FLAP DETENT					
	1	5	15	20	25	30
777-F	265	245	230	225	200	180

\*ALL FLAP PLACARDS, VCAS





# 777F Airplane Description GW/CG Envelope

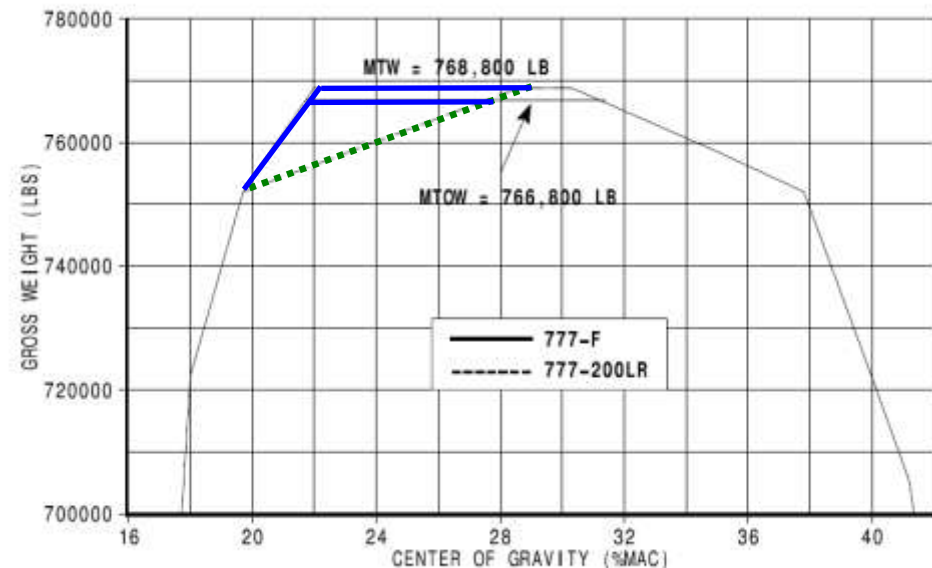


# 777F Airplane Description

## Maneuver Load Alleviation

### Maneuver Load Alleviation (MLA)

- **Purpose:** Reduce wing bending loads due to positive maneuver conditions
- **Function:** Symmetrically deflects some lateral control surfaces as a function of normal load factor
- **Benefit:** Forward CG expansion without additional structure
- **Application:**  
777F unique to-date.

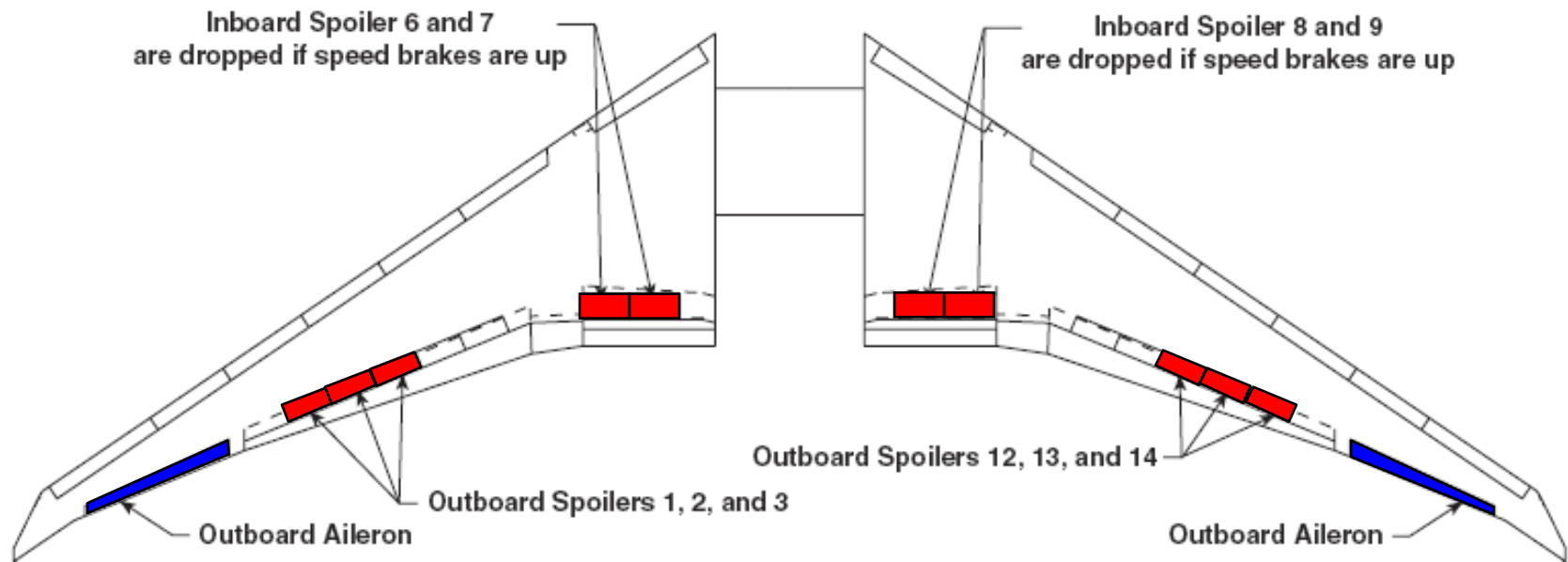


# 777F Airplane Description

## MLA Details

### Design Intent:

- Handling Qualities unchanged during normal maneuvers
- Handling Quality changes during elevated “g” maneuvers to be imperceptible



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# Early Testing...

- **First Flight: July 14<sup>th</sup>, 2008**
  - Typical Boeing first flight profile - manual column, wheel, rudder kicks.
  - No MLA operation.
  
- **Second Flight: July 16<sup>th</sup>, 2008**
  - Completed conditions not performed on first flight required to commence planned test program
  
- **Modal Stability Flight Testing: July 26-29, 2008**
  - To evaluate interactions between MLA control law and airplane structure
  - First discovered aileron vibration on 3rd flight (7/26/2008)
  
- **MLA dedicated testing and all other planned testing (smoke, etc.)**
  - Put on hold until vibration issue could be discussed and/or fixed.

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# Aileron vibration discovery flights

## ■ Flutter-type testing

- 2 altitudes and increasing Mach numbers
  - Manual controller “kicks”
  - Elevator frequency sweeps
  - Roller Coasters and “kicks” in turns (elevated g)

## ■ Aileron vibration occurred with trailing edge up deflections of ailerons.

- Ailerons had never been used previously at these speeds due to aileron lock-out function.
- With additional MLA g bias applied (e.g.- PFC thinks it's under a 1.8 G load): column kicks, shallow bank turns exhibited vibration
- Without MLA g bias: 60 deg bank turns and roller coaster maneuvers to higher g levels showed vibration

# Aileron Vibration Details

## ■ Pilot comments on Aileron Vibration:

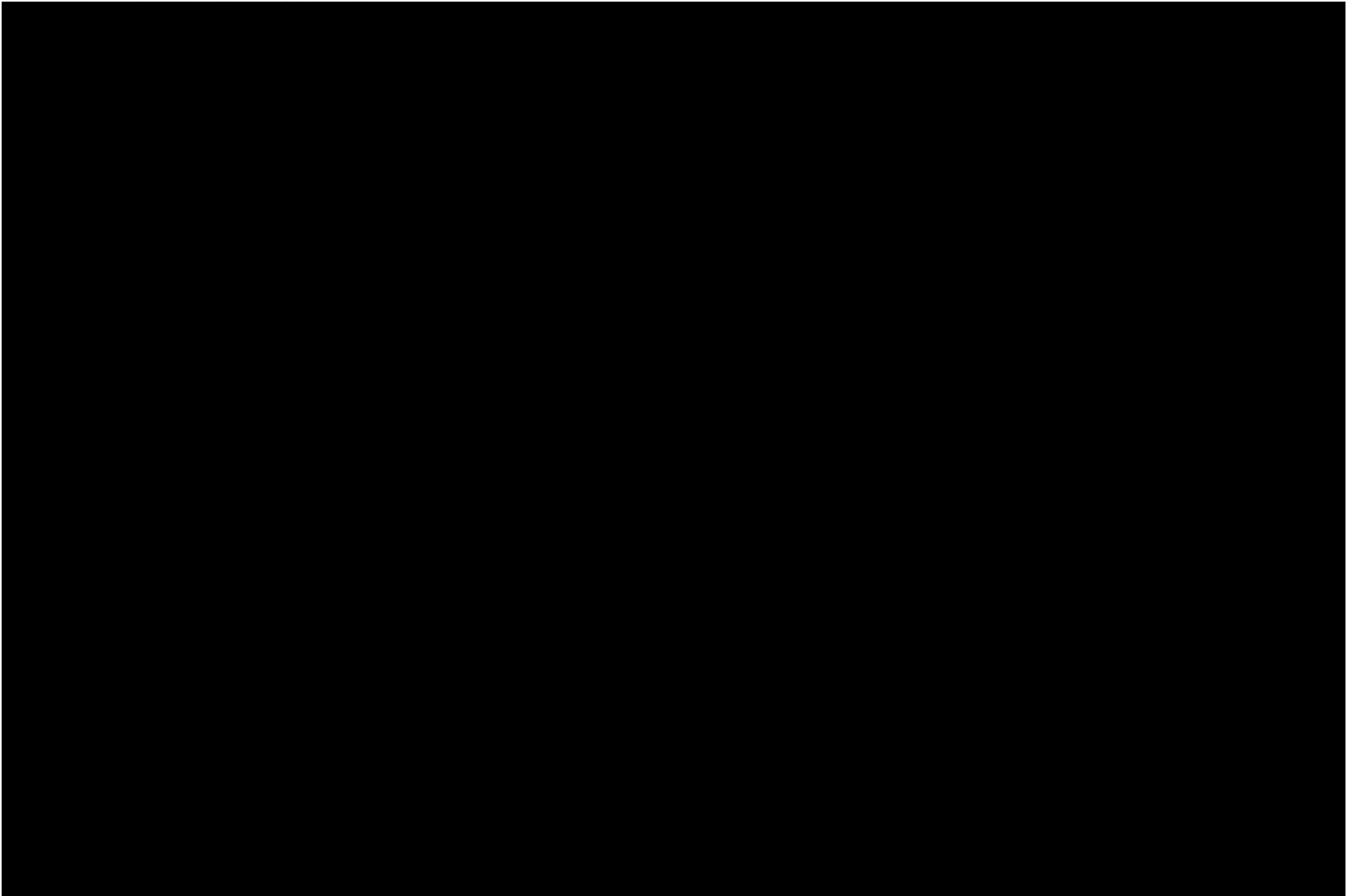
- “Alarming.”
- “Feels like driving over a washboard road.”
- “Feels like something is going to come off.”
- “Distracting and you naturally want to unload and slow down”

## ■ Causal factor unknown.

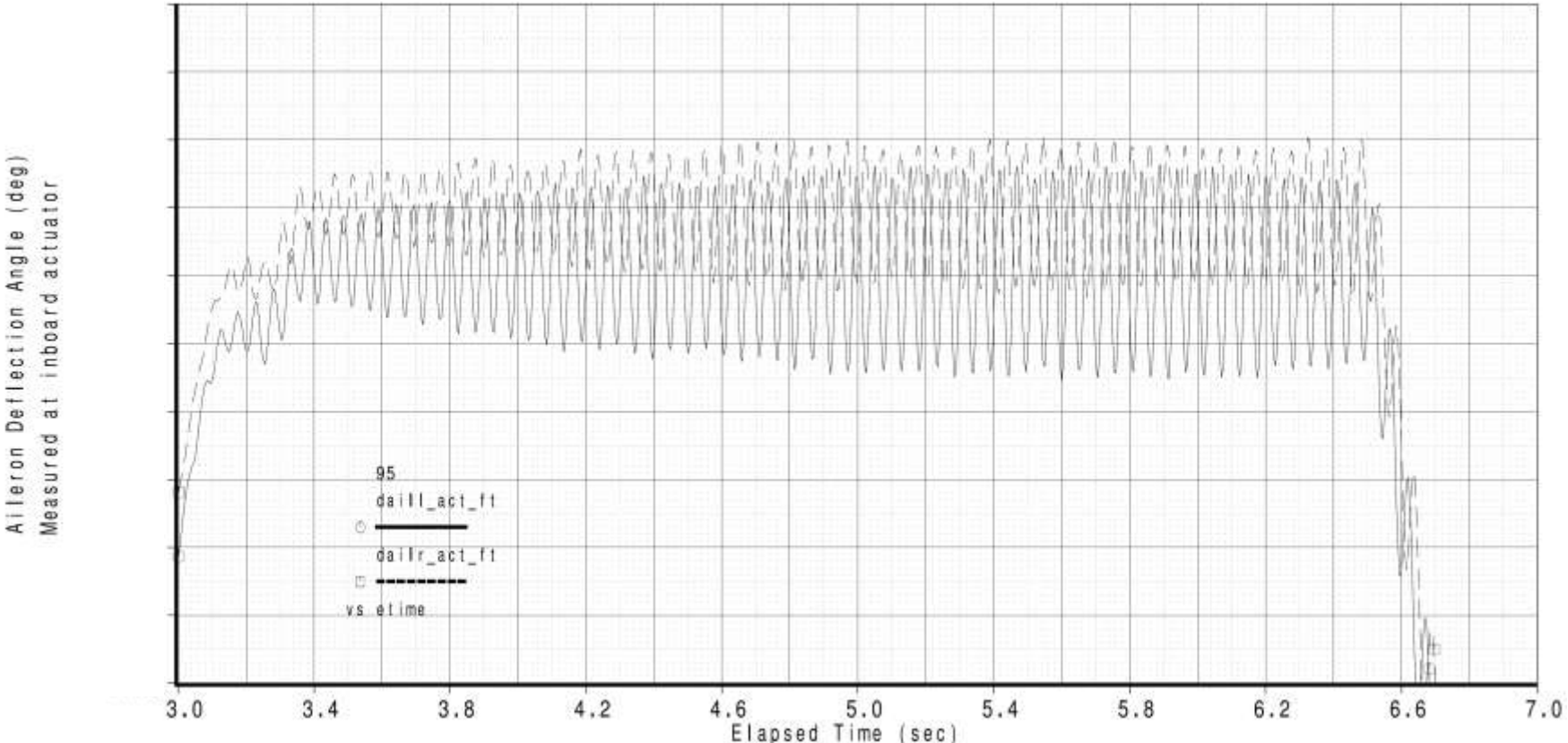
- Potential contributing factors include:
  1. Aileron deflection angle for vibration corresponds to approximately zero hinge moment
  2. A “scoop” is created between the mass balance tower and the lower surface of the wing when the aileron is deflected



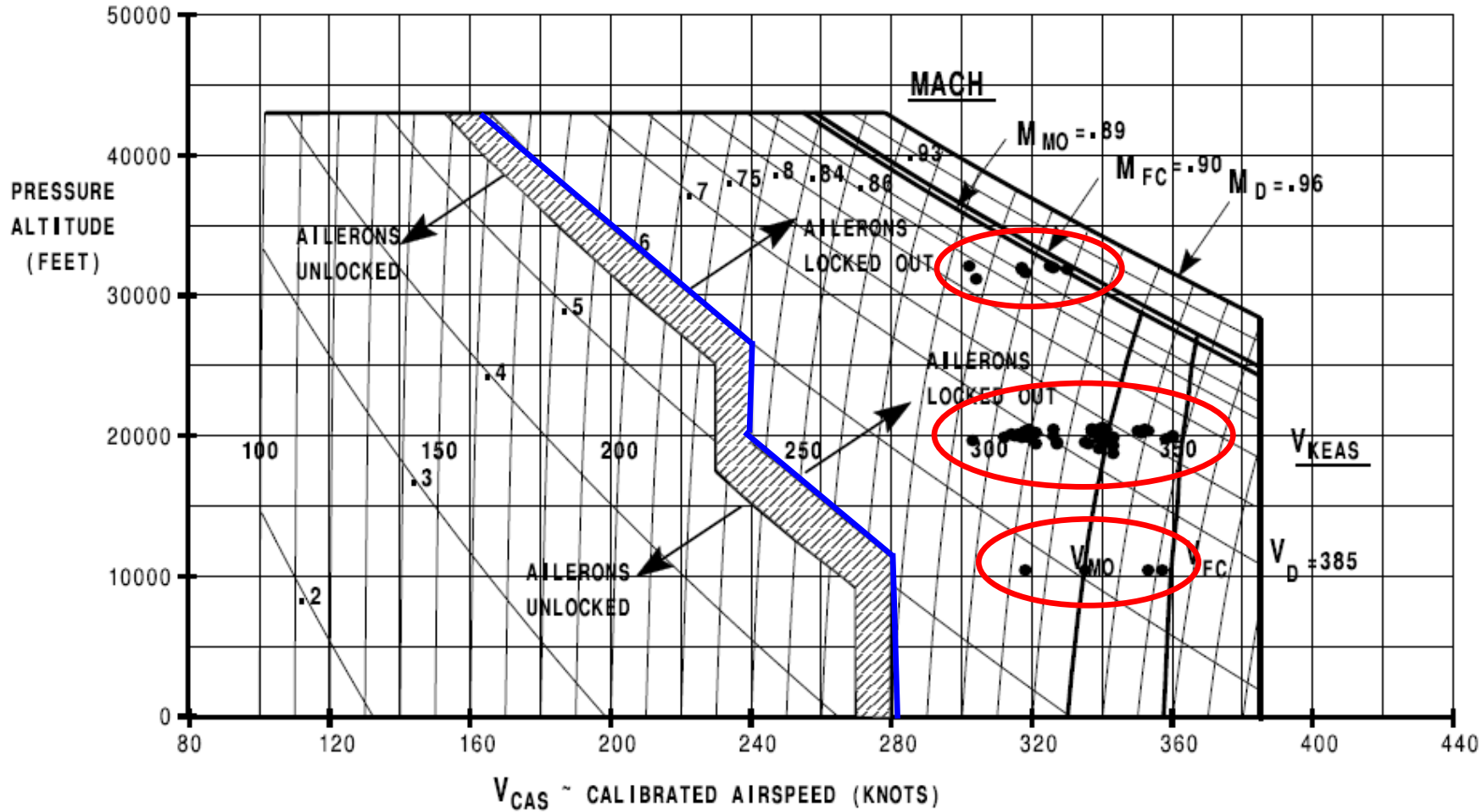
# Aileron Vibration Video



# Aileron Vibration Details



# Aileron Vibration Mapping



# 777 Outboard Aileron Balance Tower



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# To fix or not to fix ?

- **Likelihood of occurrence in service (1.9g event)**
  - Remotely encountered, if ever, in revenue service
  
- **Pilot opinion**
  - Needs to be fixed !
  - Avoidance maneuver consequences
  - Distracting (flight deck vibration cert issue)
  
- **Flutter concern**
  - No issue – mass balanced surface
  
- **Fatigue concern**
  - Further data required
  
- **Economics (Management opinion)**
  - Desire to not spend money because outside the “normal” envelope

# Potential remedies

- **Aerodynamic modifications**
  - External wing treatments
  - Aerodynamic sealing
  - Internal flow diversion
  
- **Flight control software modifications**
  - Change to basic MLA aileron travel schedule
  - Addition of aileron “No-Dwell zone” function

**Chosen solution**



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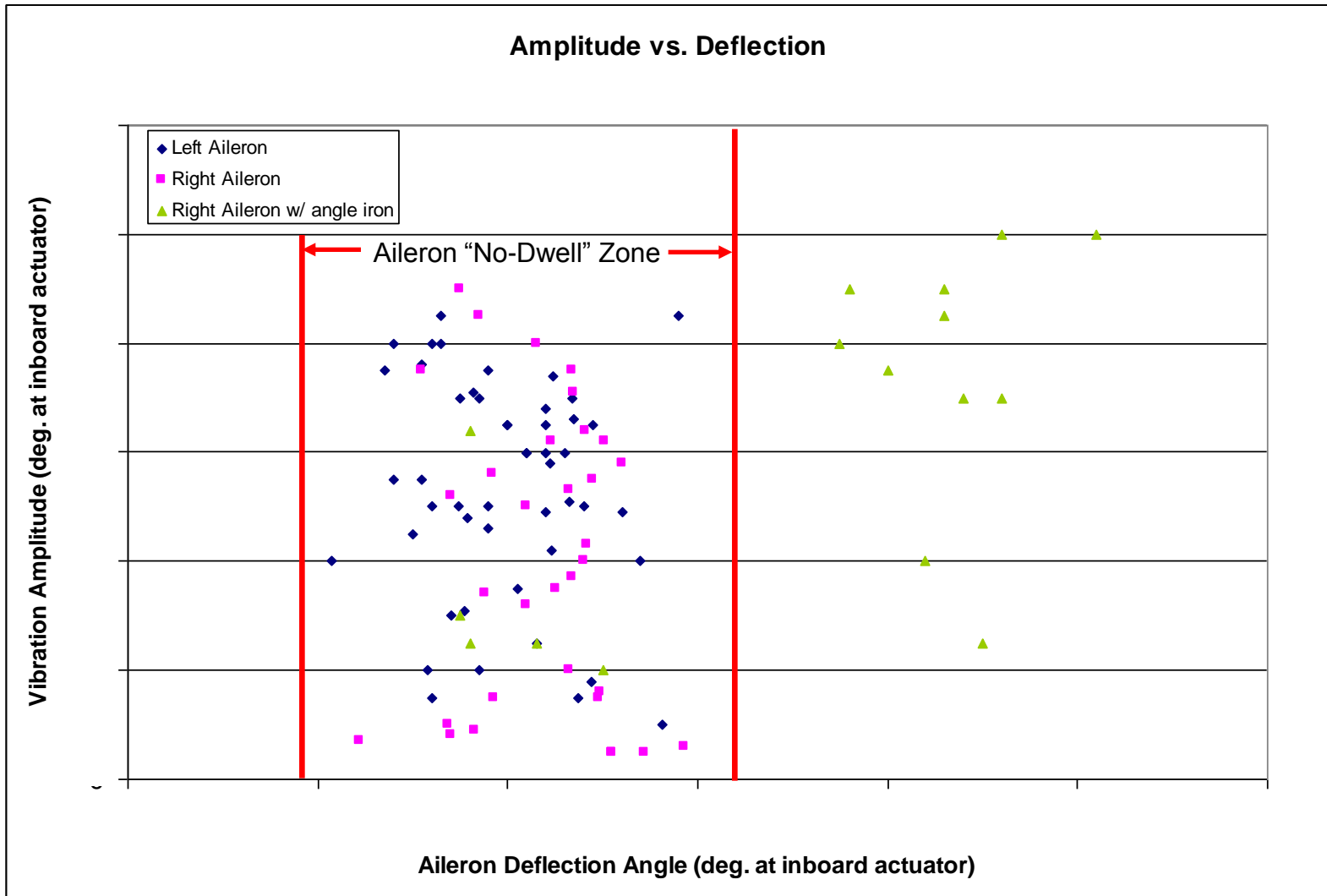
# Re-test flights : Test Plan

- **Flight test on August 6 with additional instrumentation and angle iron added**
  - Similar test conditions as before (kicks and turns)
  - Two altitudes (including one lower than previous)
  - Expected vibration and successfully repeated it
  
- **Flight test on August 18 with modified flight control law for MLA (aileron no-dwell zone functionality)**
  - Exclusively did shallow turns (with and without 0.8 g bias)
  - Tested with rollout configuration and with no-dwell active
  - Successfully eliminated vibration with no-dwell function

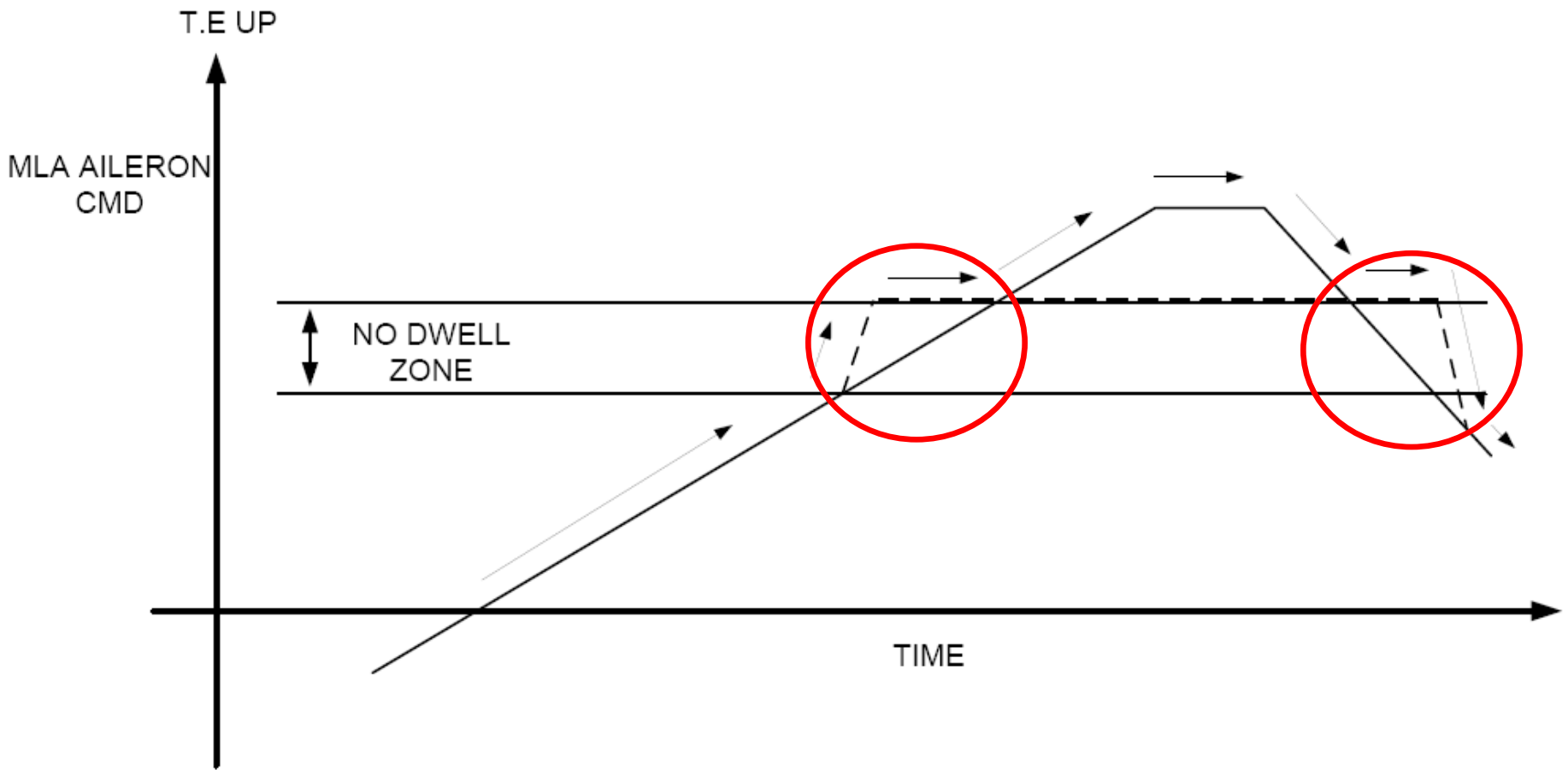
# Re-test flights : Aileron Vibration Investigation

- **New Flight Test Parts/Instrumentation installed for:**
  - **Right Wing Aileron:**
    - Angle Iron attached in front of the outboard aileron balance tower.
    - Idea: Can we affect (not eliminate) the vibration by modifying local flow?
  - **Left Wing Aileron:**
    - Kulite pressure sensor to measure pressures in the cavity of balance tower
    - Strain gauge added to outboard aileron hinge for fatigue analysis
    - Accelerometer added to outboard trailing edge for loads analysis.

# Re-test flights : Aileron Vibration Mapping



# Re-test flights : MLA Aileron "No Dwell" Zone



# 777F MLA Flight Testing (once vibration eliminated)

- **Steep Turns**
  - Determine if MLA interferes with pilot's ability to perform maneuver.
  
- **Rapid Elevator Inputs / Avoidance Maneuvers**
  - Ensure acceptable performance with MLA
  
- **Wind-Up Turns**
  - Flown at conditions where deterrent buffet would be  $\geq 2.5$  g's
  
- **Roller Coasters**
  - Evaluate handling qualities in pitch axis as MLA engages and disengages

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# Lessons Learned

- **Fly-By-Wire airplane control laws present opportunities to remedy issues such as this**
- **Team approach to problem solving is required**
  - Especially when pilot opinion and engineering data differ
- **The “right thing” will still prevail once thorough communication occurs**
- **Important to capture lessons learned on specific technical “discoveries” to share between programs**
- **Pilot distraction needs to be considered as certification issue**

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# Summary

*Never assume that a test program will escape problems, no matter how seemingly "minor" the effort is!*



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